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Dynamic Modeling and Validation of a Carbon Dioxide Heat Pump System

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Abstract: The objective of the present manuscript is to develop a dynamic model to simulate the transient performance of a heat pump system that utilizes environmental-friendly refrigerant, Carbon Dioxide (R-744). Exploiting the high customizability of the modeling tools supporting acausal, equation-based, and object-oriented modeling approach, a dynamic model for a transcritical heat pump system was developed. The model was validated with the experimental results with particular attention to the refrigerant mass flow rate, cooling/heating capacity, compressor power, and the coefficient of performance.

Keywords: dynamic modeling; validation; carbon dioxide (R-744); heat pump system;

1. Introduction

Smart society and connected devices can be considered as a new technology era. With the forthcoming 5G technology, ubiquitous smart-devices offer tremendous opportunities for energy optimization and efficiency gain. One of the main features of the smart society is operating the connected devices which smartly controlled based on real-time demand¹⁾. From this perspective, it is important to understand the transient behaviors of the devices while heating, ventilation, air-conditioning, and refrigeration (HVAC&R) systems constitute a dominating share of energy usage. Thus, chillers and heat pumps can no longer be considered as the devices that operate at steady-state.

Table 1. Transient simulation tools.

Tool	Modeling approach
TRNSYS ²⁾	Causal, non-equation-based, non-object-oriented
Modelica ³⁾ /Dymola ⁴⁾	Acausal, equation-based, object-oriented
MATLAB/Simulink ⁵⁾	Causal, non-equation-based, object-oriented
MATLAB/Simscape ⁶⁾	Acausal, equation-based, object-oriented

The preliminary design of a smart-controlled thermal system is one of the critical phases in its development. In most cases, the initial design is developed through

numerical and analytical treatments. Modeling and simulation offer huge advantages as compared to the experimental procedure. However, developing a transient simulation model for a complex system is often a time-consuming and effort-demanding process. Knowledge of building up a network connection and integration of all physical parameters in submodels is essential. Additionally, a proper ordinary differential equation (ODE) solver needs to be selected to avoid some numerical hiccups during simulations.

These challenging tasks have been resolved by the help of simulation programming tools that offer different paradigms of modeling approaches; therefore, engineers could put more effort into physical modeling. Transient simulation tools, which have been frequently employed for the research on thermal systems, are categorized in Table 1. According to the previous studies which compared the acausal, equation-based, and object-oriented tool and the causal, non-equation-based, and non-object-oriented tool^{7),8),9)}, it was revealed that the former is a more effective method than the latter for creating user-defined models to implement of model components; the latter was cited as burdens to engineers in creating a new component model. Due to the disadvantages of the simulation tools based on the causal and non-equation-based approach as well as with the increasing demands of complex thermal systems, causal and non-equation-based approach modeling tools are often shunned by the industry. This void is filled up by acausal, equation-based, and

object-oriented modeling tools that offer a multi-physical domain. Modelica, as a well-known and open-source tool, has already passed numerous experimental validations. On the other hand, Simscape has merits in terms of generating fewer equations and it is not only powerful and user-friendly for control purposes but also can provide more flexible and easy ways for users to extend their models into a multi-physical domain by offering an integrated-tool platform¹⁰⁾; nevertheless, the reported research works on the modeling and simulation of physical systems using Simscape are considered rare^{11),12),13)}. Recently, Singh and Sørensen¹⁴⁾ developed a dynamic simulation model based on Simscape for an R-407C heat pump system. Using the model, they investigated the effects of ambient temperature on the system performance and room temperature. However, the validation was absent.

Meanwhile, Chlorofluorocarbon (CFC) and Hydrochlorofluorocarbon (HCFC) have been dominantly used as the working fluids in the conventional vapor compression systems. However, they are on the phase-out process because of their high ozone-depleting potential (ODP) and global warming potential (GWP). Carbon dioxide (R-744) is one of the promising alternative refrigerants with zero ODP and GWP of one¹⁵⁾. R-744 has a low critical temperature of 31.1 °C so, in most cases, R-744 heat pump systems are operated by transcritical cycles. In the transcritical cycle, heat rejection at the high-pressure side occurs not by the condensation, but by the gas cooling. During the gas cooling process, refrigerant temperatures and pressures are independent of each other; moreover, near the pseudocritical line, there is a dramatic change in the thermophysical properties of R-744. These facts become challenges for engineers to modeling R-744 systems.

Over the last couple of decades, various researches have been conducted, empirically and numerically, for R-744 heat pump systems to investigate their performance and improve it. To understand the transient behavior of R-744 heat pump systems and their performance, researchers have tried to develop dynamic models. Pfafferott and Schmitz¹⁶⁾ developed a library for R-744 refrigeration systems for aircraft in Modelica. By utilizing that library, they developed a simulation model and validated their model with experimental results. Shi et al.¹⁷⁾ developed a dynamic model of an R-744 supermarket refrigeration system using Dymola. They compared the simulation results to the field data. Zheng et al.^{18), 19)} developed their dynamic simulation model for an R-744 refrigeration system equipped with an ejector and improved it for a two-stage evaporating system. They validated their model by focusing on the ejector performance and corresponding system behavior. Bush et al.²⁰⁾ built up an R-744 transcritical refrigeration system for supermarket applications using the pre-validated component model

based on Modelica/Dymola. After the system model was experimentally validated, the system's response was investigated when the system encounters sudden changes in thermal load.

To develop smart HVAC&R systems based on real-time demands, it is essential to understand the dynamic behaviors of the connected devices. In this perspective, well-validated dynamic models are considered as useful tools for engineers because they enable to investigate the system performance at various operating conditions and to simulate control methods, minimizing the extensive experimental procedures. Several dynamic models for R-744 transcritical heat pump systems were reported in the literature, however, when it comes to being compared to conventional system models that are operated by subcritical cycles, transcritical models of R-744 systems are rather limited. The objective of the present study is to develop and simulate the transient behavior of an R-744 transcritical heat pump system. The model meticulously attempts to incorporate the heat transfer mechanisms in the evaporator and the gas cooler as well as detailed models for the compressor and the expansion process. The Simscape simulation environment was adopted utilizing the object-oriented and customizable technique. The dynamic response of the R-744 heat pump system is validated using in-house experimental data.

2. Methodology

2.1 Actual heat pump system

A schematic of the experimental apparatus is shown in Fig. 1 (a). The apparatus consists of an R-744 heat pump system, a cooling water loop, and a heating water loop. The heat pump consists of a scroll compressor, which is controlled by an inverter, a tube-in-tube type heat exchanger (HX) as a gas cooler, an electronic expansion valve (EEV), and a plate-type HX as an evaporator. The schematic of the gas cooler and the evaporator are shown in Fig. 1 (b). The gas cooler consists of three inner-tubes and a single outer-tube. The refrigerant flows inside the three inner-tubes while the cooling water flows annular space between the inner-tubes and outer-tube, in counter-current direction to the refrigerant flow. The evaporator consists of ten channels of the refrigerant flow and eleven channels of the heating water flow. Specification of the mentioned four main components of the heat pump system is summarized in Table 2. An oil separator at the compressor discharge is utilized to rid the effects of lubricant oil circulation. Sight-glasses, at the evaporator inlet and the compressor suction, are to check the refrigerant flow phase. Each water loop consists of a constant temperature water bath, a circulation pump, a gear-type flow meter, and a by-pass valve.

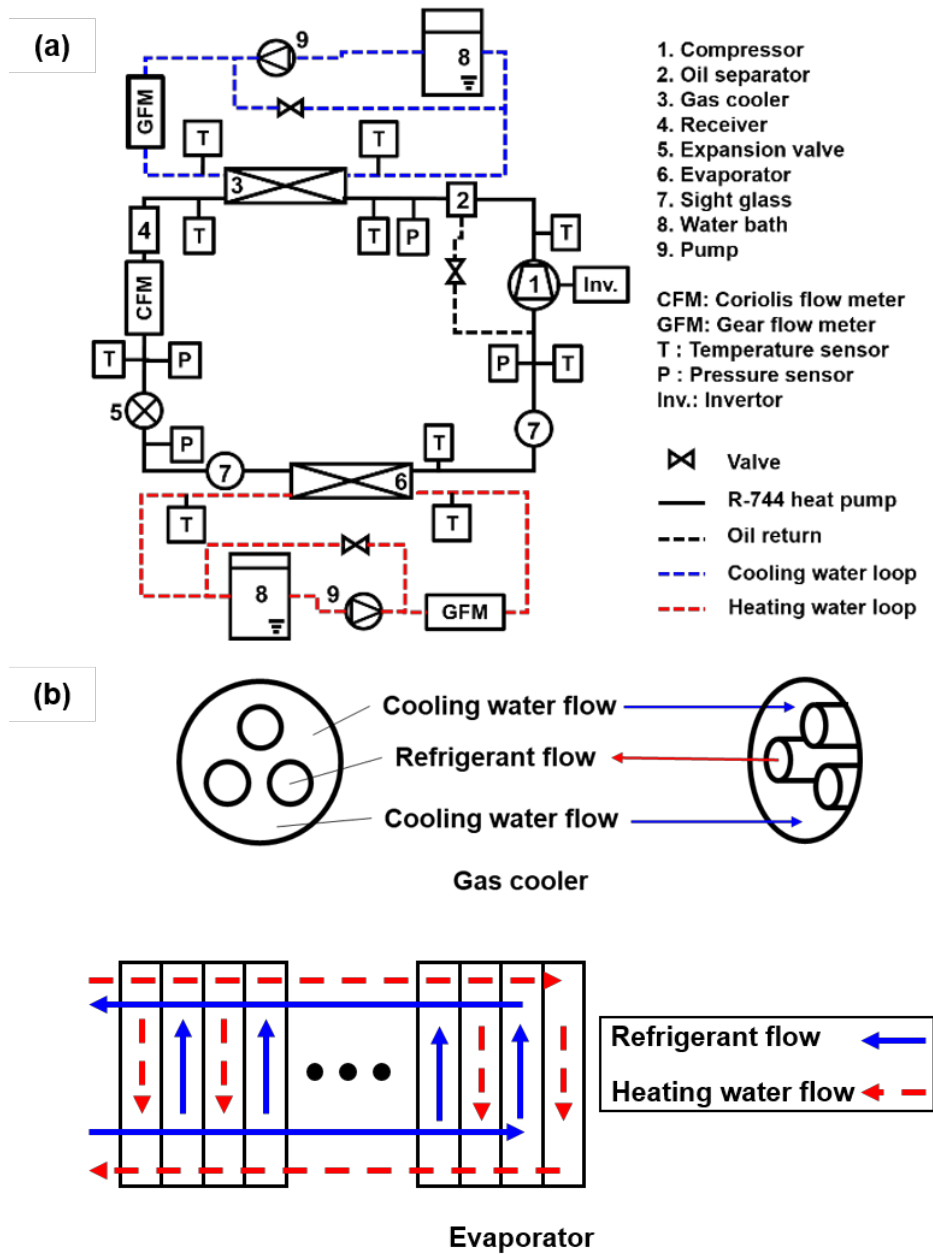


Fig. 1: (a) schematic of experimental apparatus, (b) schematic of the gas cooler and the evaporator.

During the experiments, pressure and temperature of the refrigerant flow were measured at each main component's inlet and outlet by pressure sensors and K-type thermocouples, respectively. The refrigerant mass flow rate was measured by a Coriolis mass flow meter installed at the inlet of the EEV. Temperatures of the water at the inlet and outlet of each HX were measured by K-type thermocouples. The water flow rate in each water loop was measured by the gear-type flow meter. The flow was regulated by the by-pass valve. Controlling the water temperature at each HX inlet was performed by the constant temperature water bath installed in each water loop. In the gas cooler, the refrigerant heat was rejected to the cooling water, while in the evaporator, the heating

water heat was absorbed into the refrigerant. The degree of superheating at the compressor suction was adjusted by the opening step of the EEV. Electric signal using a digital panel manually controlled the EEV's opening step to adjust the degree of suction superheating. The rotation speed of the compressor was controlled by the inverter. The compressor and inverter are connected to a power meter. Readings of power and rotation speed were indicated in a display window to check its input power and speed. All the measuring instruments are connected to a data acquisition system. Their readings were recorded every second. The experimental range carried out in the present study is tabulated in Table 3.

Table 2. Specification of the experimental heat pump.

Component	Content	Value or range	Unit
Compressor	Swept volume	4.0	cm ³
	Rotation speed range	30 to 90	rev/s
	Diameter of suction and discharge tubes	6.53	mm
Gas cooler	Inner diameter of the inner tubes	2.8	mm
	Outer diameter of the inner tubes	4.4	
	Inner diameter of the outer tube	14.6	
	Outer diameter of the outer tube	16.0	
	Tube length	13.6	m
	Overall heat transfer area	0.578	m ²
EEV	Diameter of inlet and outlet tubes	6.35	mm
	Maximum pressure at the inlet	14	MPa
	Maximum differential pressure	10	
	Operating temperature range	-30 to 70	°C
	Opening step (close – full open)	0 to 500	-
Evaporator	Width	90	mm
	Depth	35	
	Height	434	
	Overall heat transfer area	0.648	m ²

Table 3. Experimental range.

Operating parameter	Range	Unit
Gas cooler		
Water flow rate	45 to 145	L/h
Water inlet temperature	20 to 30	°C
Evaporator		
Water flow rate	77 to 350	L/h
Water inlet temperature	15 to 20	°C
Charge	1.4 to 1.7	kg
Refrigerant mass flow rate	35 to 55	kg/h
Suction pressure	3.4 to 4.7	MPa
Discharge pressure	7.8 to 10.5	
Suction superheating	2 to 20	°C
Discharge temperature	70 to 100	

Based on the measured data, the heat transfer rates in the refrigerant-side (\dot{Q}_r) and water-side (\dot{Q}_w) at each HX were calculated by:

$$\dot{Q}_r = \dot{m}_r \Delta h_{r,i,o}, \quad (1)$$

$$\dot{Q}_w = \dot{V}_w \rho_{w,i} \left(\frac{C_{p,w,i} + C_{p,w,o}}{2} \right) \Delta T_{w,i,o}, \quad (2)$$

where $\Delta h_{r,i,o}$, \dot{V}_w , $\rho_{w,i}$, $C_{p,w,i}$, $C_{p,w,o}$, and $\Delta T_{w,i,o}$ represent the refrigerant specific enthalpy difference between the inlet and outlet of the HX, the water flow rate, the water density at the inlet of the HX, the specific heats of water at the inlet and outlet of the HX and the temperature difference of water between at the inlet and outlet of the HX, respectively. All thermodynamic properties of the refrigerant and water were obtained from Lemmon et al.²¹⁾.

Fig. 2 shows the heat balance test results at each HX in steady-state conditions. All experimental data is within

10% of heat balance error, and 90% of total data shows a good agreement within a 6% error. Coefficient of performance (COP) for cooling and heating was assessed by:

$$COP_c = \frac{\dot{Q}_{w,evap}}{\dot{W}}, \quad (3)$$

$$COP_h = \frac{\dot{Q}_{w,gc}}{\dot{W}}, \quad (4)$$

where $\dot{Q}_{w,evap}$ and $\dot{Q}_{w,gc}$ denote the heat transfer rates of the water-side for evaporator and gas cooler, respectively.

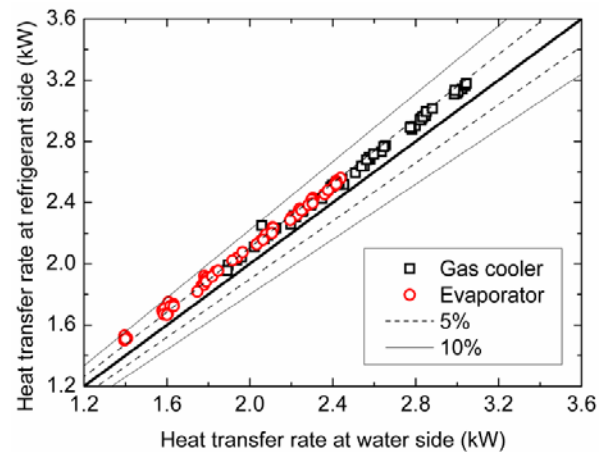


Fig. 2: Experimental heat balance in HXs.

The reduced data, such as heat exchange rate at each HX and COPs, have a relationship with measured data (i.e., independent variables) as follows:

$$\Psi = f(\psi_1, \dots, \psi_n), \quad (5)$$

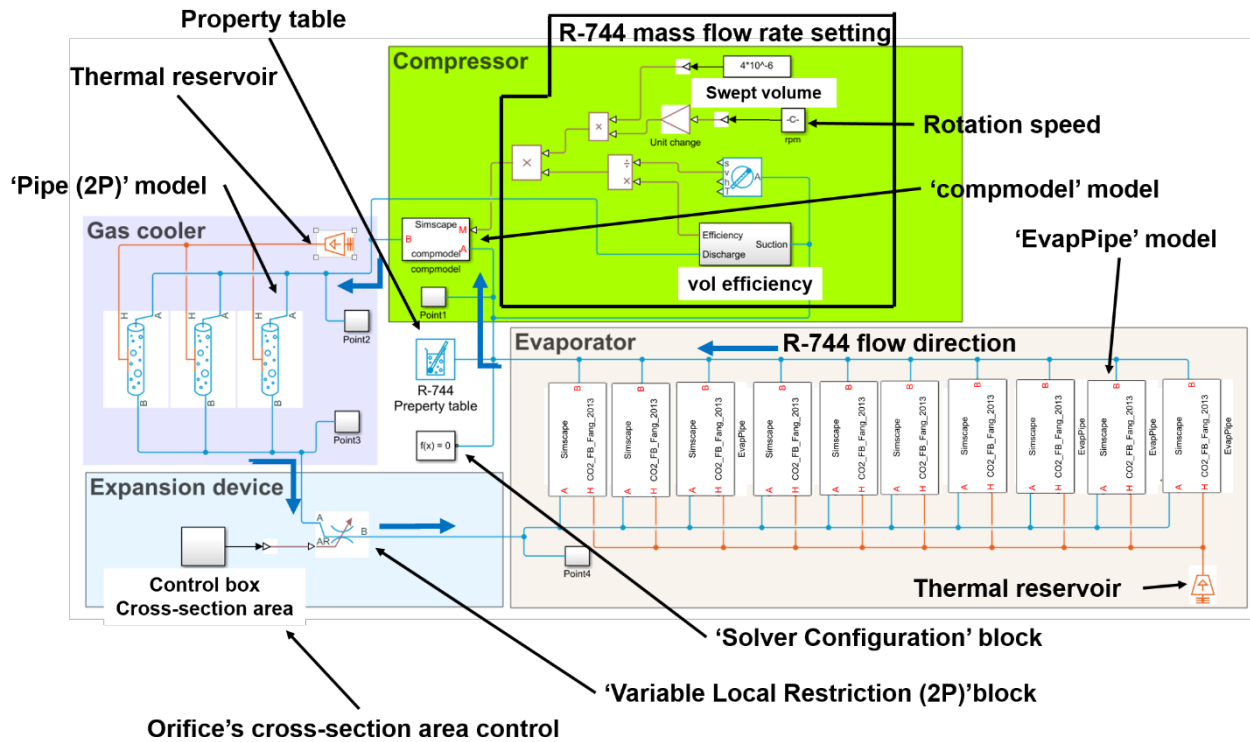
where Ψ and ψ represent the reduced data and the measured data, correspondingly. Uncertainty of the reduced data can be calculated by:

$$U_{\Psi} = \left(\sum_{i=1}^n \left(\frac{\partial}{\partial \psi_i} f(\psi_1, \dots, \psi_n) \right) U_{\psi_i} \right)^{0.5}, \quad (6)$$

where U_{Ψ} and U_{ψ_i} represent the uncertainty of the reduced data and the uncertainty of independently measured data, respectively. Uncertainties of measured and reduced data are summarized in Table 4.

Table 4. Uncertainties of measured and reduced data.

Data type	Content	Uncertainty	Unit
Measured data	Temperature	± 0.1	$^{\circ}\text{C}$
	Pressure		
	at the gas cooler inlet	± 0.044	MPa
	at the EEV inlet	± 0.040	
	at the EEV outlet	± 0.015	
	at the suction	± 0.017	
	Refrigerant mass flow rate	$\pm 1.08\%$ of reading	kg/h
	Water flow rate		
	at the cooling water loop	± 1.0	L/h
	at the heating water loop		
	Compressor power	$\pm (0.1\% \text{ of reading} + 0.002)$	kW
Reduced data	Heating capacity of the gas cooler	$\pm 5.4\%$ of the value	kW
	Cooling capacity of the evaporator	$\pm 5.3\%$ of the value	
	COP_h	$\pm 5.5\%$ of the value	
	COP_c	$\pm 5.4\%$ of the value	



(a) Model diagram

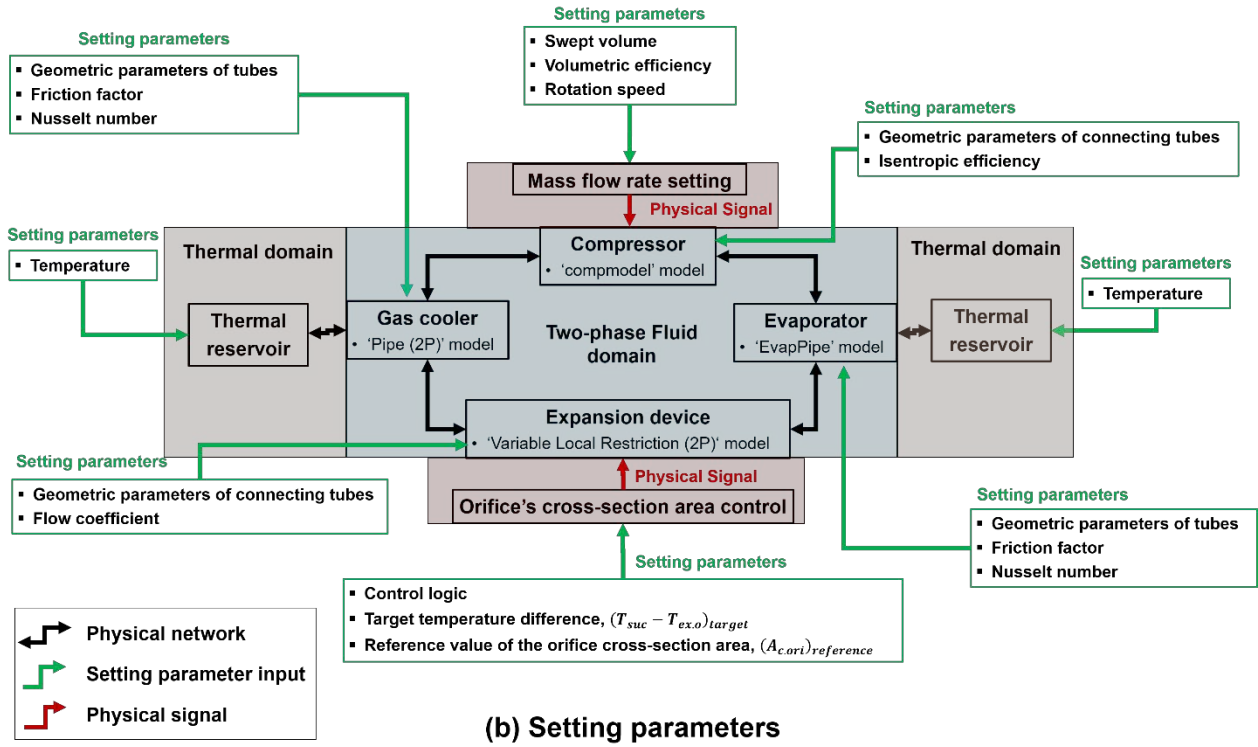
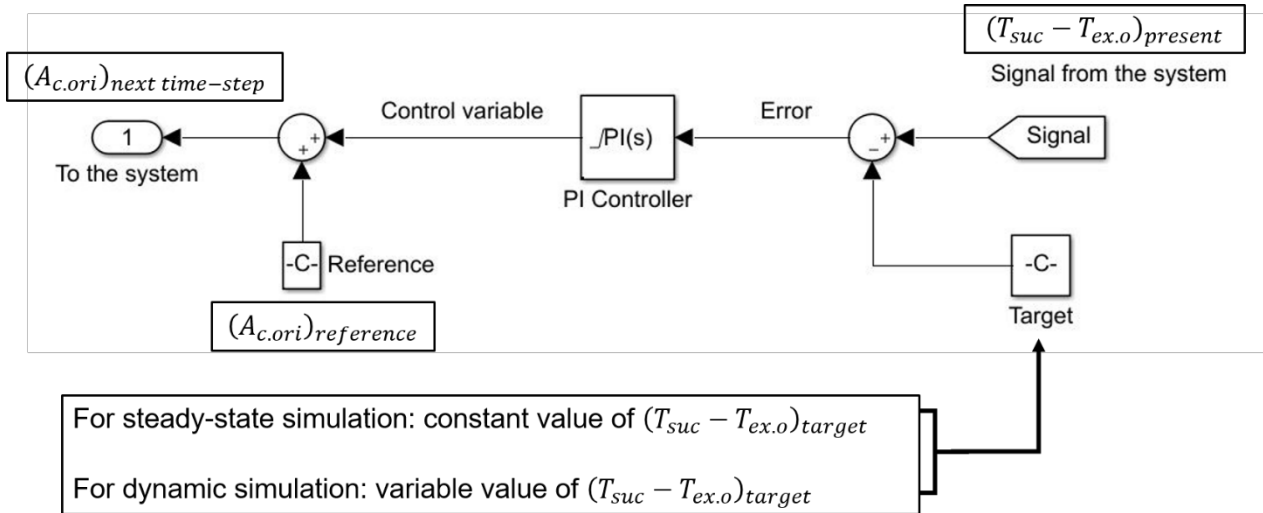


Fig. 3: (a) system model in the Simscape™/MATLAB, (b) setting parameters.



2.2 System model description

Fig. 3 (a) shows the heat pump system model. The system consists of four subsystems; a compressor, a gas cooler, an expansion device, and an evaporator. Each subsystem represents each main component of the experimental heat pump. Fig. 3 (b) describes the setting parameters for each subsystem model and indicates the physical network.

In Fig. 3 (a), the compressor model consists of the 'compmodel' block which is the user-defined model. The model's geometric parameters, such as the swept volume and diameters of connecting tubes, were set up with the identical values of the experimental compressor. The

black box named 'R-744 mass flow rate setting' indicates the refrigeration mass flow rate calculation. The setting value of the mass flow rate is calculated based on the input rotation speed, the refrigerant density at the suction, the swept volume, and the volumetric efficiency. The calculated mass flow rate then becomes the input of the 'compmodel' block. In the black box, a subsystem named 'vol efficiency' represents volumetric efficiency calculation. The consisting equations for the 'compmodel', 'R-744 mass flow rate setting', and 'vol efficiency' will be mentioned in Section 2.3.

In the gas cooler model, three 'Pipe (2P)' blocks represent three inner-tubes in the experimental gas cooler.

The hydraulic diameter and the length of the ‘Pipe (2P)’ blocks are 2.8 mm and 13.6 m, respectively, which are identical with that of the experimental gas cooler. The evaporator model consists of ten ‘EvapPipe’ blocks which represent ten channels of the refrigerant flow in the experimental evaporator. In the model, the hydraulic diameter of ‘EvapPipe’ was assumed 3 mm; thereby the length of ‘EvapPipe’ was determined to 6.9 m based on the heat transfer area and the number of refrigerant flow channels in the experimental evaporator. In the gas cooler and evaporator, all the ‘Pipe (2P)’ and ‘EvapPipe’ blocks are connected to each temperature setting block that represents thermal reservoirs. When the refrigerant flows into each ‘Pipe (2P)’ and ‘EvapPipe’, the flow is uniformly distributed in each ‘Pipe (2P)’ and ‘EvapPipe’.

The expansion device model consists of the ‘Variable Local Restriction (2P)’ block and the ‘Control box Cross-section area’ block. The ‘Variable Local Restriction (2P)’ block represents an orifice. The diameters of tubes in which connecting to the ‘Variable Local Restriction (2P)’ model were set up with the identical values of the experimental EEV. The cross-section area of the orifice is controlled at the ‘Control box Cross-section area’ block. Fig. 4 illustrates the control logic of the expansion device. The target temperature difference between the suction and the expansion device outlet, $(T_{suc}-T_{ex.o})_{target}$, becomes the control ‘Target’. The temperature difference at the present step, $(T_{suc}-T_{ex.o})_{present}$, represents the ‘Signal’ from the system. The ‘Control variable’ is calculated by PI-method. The proportional and integral coefficients of the PI-method were obtained by the linearization approach using ‘PID Tuner’ supported by the Simulink. The cross-section area of the orifice for the next time-step, $(A_{c.ori})_{next\ time-step}$, was calculated based on the reference value, $(A_{c.ori})_{reference}$, considering the obtained ‘Control variable’.

The four subsystems are connected with the blue lines representing the ‘Two-Phase Fluid (2P)’ domain. During the simulations, a set of physical variables of the refrigerant, which are computed in each physical block, transfers to the adjacent physical blocks and affect each other. The orange lines represent the ‘Thermal’ domain where heat transfer happens. In the ‘Thermal’ domain, the heat transports from high temperature to low temperature based on physical law. The maroon lines in the compressor model represent the ‘Physical Signals’ domain. The black lines represent the ‘Simulink’ domain. Both ‘Physical signals’ and ‘Simulink’ domains indicate signal networks. However, signals in the ‘Physical signals’ domain are with physical units while the signals in the ‘Simulink’ domain are without physical units, it contains numbers. The blocks, namely ‘Point 1 to 4’, check the refrigerant states at each position and display their property values.

A table of R-744 properties obtained from Lemmon et al.²¹⁾ is set up in the MATLAB environment. Properties on this table are parameterized by pressure and normalized the specific internal energy. This table is brought into the Simscape’s physical domain using the ‘Two-Phase Fluid

Properties (2P)’ block. Properties that are not directly specified in the table are going to be determined by curve-fitted interpolation during the simulation. A property of the refrigerant in the saturated state, i.e., the mixture of the saturated-liquid and the saturated-vapor (ζ_2) is determined by weighting the saturated vapor quality, linearly, as follows:

$$\zeta_2 = (1 - x)\zeta_L + x\zeta_V, \quad (7)$$

where x represents the vapor quality. The subscripts L and V indicate the saturated liquid state and the saturated vapor state, respectively.

While, in an actual system, a compressor lubricant oil has a significant effect on system performance. In the present modeling, all the effect due to the presence of lubricant oil is not taken into account.

2.3 Component model description

In the following subsections, basic mathematical equations in each physical block model are briefly described based on Simscape reference²²⁾.

2.3.1 Compressor

According to Winkler (2009)²³⁾, in the transient simulation of a vapor compression system, the compressor is treated as a quasi steady-state component in most cases that the calculated mass flow rate and discharge temperature instantaneously update based on the current suction conditions and discharge pressure. This assumption is reasonable because the timescales associated with the variation of the compressor mass flow rate are very small compared to timescales associated with heat exchanger performance and refrigerant charge distribution. Also, dynamic modeling of a compressor with considering compressor motor needs an expensive computation cost and it requires very small of time steps to be included in a system-level simulation²⁴⁾. In this circumstance, there are two promising options in the compressor modeling method utilized in a system simulation; one is the map-based modeling and the other is efficiency-based modeling. A map-based model is based on manufacturer data, however, the data were obtained under steady-state conditions. Hence, the map-based model is valid only within a limited range. Besides, it is hard to take into consideration all the transient phenomena that probably influence the compressor performance during transient periods. While efficiency-based modeling has a disadvantage that it requires more effort to develop a model than using the map-based modeling approach. However, after tuning procedure, efficiency-based models can follow up with dynamic behavior as well as match actual systems. Due to this reason, the present compressor modeling adopted the efficiency-based modeling approach.

In Fig. 3 (a), ‘compmodel’ operates on the following

assumptions:

1. The compressor is treated as a quasi steady-state component.
2. The pressure drop of the refrigerant flow at the suction and the discharge connecting tube is neglected.
3. Heat gain or loss of the compressor's body is neglected.
4. The compression process of the refrigerant is isentropic.
5. Mechanical efficiency is neglected.

Mass conservation equation is as follows:

$$\dot{m}_{suc} = \dot{m}_{dis}, \quad (8)$$

where, \dot{m} represents the refrigerant mass flow rate. The subscripts suc and dis indicate the suction and the discharge, respective, The refrigerant mass flow rate is calculated by:

$$\dot{m}_{suc} = \eta_v \rho_{suc} V_{comp} S, \quad (9)$$

where η_v , ρ_{suc} , V_{comp} , and S represent the volumetric efficiency, the refrigerant density at the suction, the compressor swept volume, and the compressor rotation speed, respectively. Here, the Eq. (9) is applied to the 'R-744 mass flow rate setting' in Fig. 3 (a). Compressor power is calculated by:

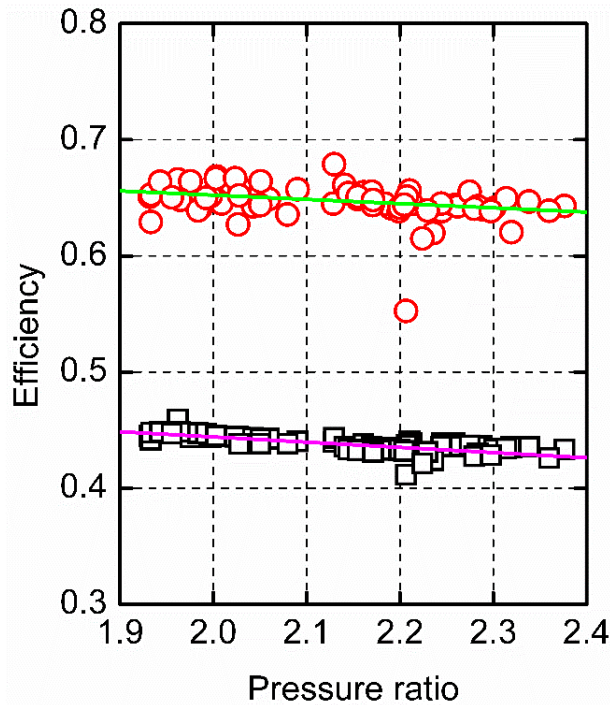
$$\dot{W} = \left[\dot{m} \left(h + \frac{(Gv)^2}{2} \right) \right]_{dis} - \left[\dot{m} \left(h + \frac{(Gv)^2}{2} \right) \right]_{suc}, \quad (10)$$

where h , G , v , and \dot{W} represent the specific enthalpy of the refrigerant, the refrigerant mass flux, the specific volume of the refrigerant, and the compressor power, respectively. The specific enthalpy of the refrigerant at the discharge is given by:

$$h_{dis} = \frac{h_{dis,s} - h_{suc}}{\eta_s} + h_{suc}, \quad (11)$$

where $h_{dis,s}$ and η_s represent the specific enthalpy of the refrigerant at the discharge by isentropic compression and the isentropic efficiency, respectively.

The volumetric efficiency (η_v) and isentropic efficiency (η_s) were derived from a set of 63 experimental data obtained under each steady-state condition independently, using the experimental heat pump system mentioned in Section 2.1. The experimental data are depicted in Fig. 5. The pressure ratio in the horizontal axis represents the ratio of the discharge pressure based on the suction pressure. In Fig. 5, the experimental volumetric efficiency and the experimental isentropic efficiency were computed from the following equations:



Experimental results (Number of data: 63)

- Volumetric
- Isentropic

Fitting equations

$$\eta_v = 0.5341 - 0.045(P_{dis}/P_{suc})$$

*AD = 0.04%, **MD = 0.81%

$$\eta_s = 0.7398 - 0.0438(P_{dis}/P_{suc})$$

*AD = 0.26%, **MD = 1.47%

* Average deviation

** Mean deviation

Fig. 5: Experimental data of the compressor efficiency.

$$\eta_{v.exp} = \left(\frac{\dot{m}_{r.exp}}{\rho_{suc} V_{comp} S} \right), \quad (12)$$

$$\eta_{s.exp} = \left(\frac{h_{dis.s} - h_{suc}}{h_{dis} - h_{suc}} \right). \quad (13)$$

Based on the experimental efficiencies, the fitting equations were derived as follows:

$$\eta_{v.pred} = 0.5341 - 0.045 \left(\frac{P_{dis}}{P_{suc}} \right), \quad (14)$$

$$\eta_{s.pred} = 0.7398 - 0.0438 \left(\frac{P_{dis}}{P_{suc}} \right). \quad (15)$$

The obtained fitting equations are applied to the compressor model. The Eq. (14) is applied to the ‘vol efficiency’ in Fig. 3 (a). The Eq. (15) is stated in the source code of ‘compmodel’.

2.3.2 Gas cooler and evaporator

The gas cooler and evaporator models were constructed based on the following assumptions:

1. Refrigerant flow is one dimensional and fully developed.
2. The pressure drop of the refrigerant flow at the inlet and the outlet of each tube is neglected.
3. The gravitational effect of the refrigerant flow is neglected.
4. The liquid-vapor mixture flow is treated as a homogeneous flow.
5. The effect of HX’s mass and its thermal conduction are neglected.
6. The second-side working fluid of each HX is considered a stagnant thermal reservoir.
7. The heat transfer rate is calculated concerning the refrigerant bulk temperature at the center of each tube (Fig. 6).

Based on the above assumptions, the ‘Pipe (2P)’ model and ‘EvapPipe’ model calculate the refrigerant heat transfer rate and pressure drop base on conservation equations. Fig. 6 shows the segment for mass, energy and pressure drop calculation.

Mass and energy conservation equations are calculated based on the segment described in Fig. 6 (a). A single tube is treated as a unit segment. Mass conservation is as follows:

$$\left[\left(\frac{\partial \rho_{cen}}{\partial P_{cen}} \right)_{u_{cen}} \left(\frac{dP_{cen}}{dt} \right) + \left(\frac{\partial \rho_{cen}}{\partial u_{cen}} \right)_{P_{cen}} \left(\frac{du_{cen}}{dt} \right) \right] V =$$

$$\dot{m}_o + \dot{m}_i, \quad (16)$$

where ρ , P , u , t , and V represent the refrigerant density, the refrigerant pressure, the specific internal energy of the refrigerant, the time and the inner volume of the tube, respectively. The subscripts i , o , and cen indicate the inlet of the tube, the outlet of the tube, and the center of the refrigerant volume inside the tube. Energy conservation equation is given as follows:

$$\left[(\dot{m}_i + \dot{m}_o) \left(\frac{du_{cen}}{dt} \right) + u_{cen} \left(\frac{d\dot{m}_i}{dt} + \frac{d\dot{m}_o}{dt} \right) \right] = \left[\dot{m} \left(h + \frac{(Gv)^2}{2} \right) \right]_o + \left[\dot{m} \left(h + \frac{(Gv)^2}{2} \right) \right]_i + \dot{Q}, \quad (17)$$

where \dot{Q} represents the heat transfer rate of the refrigerant flow. Here, based on the mentioned assumptions 5, 6, and 7, the heat transfer rate can be calculated as follows:

$$\dot{Q} \cong \lambda_r A (T_{r.cen} - T_{res}), \quad (18)$$

where λ_r , A , $T_{ref.cen}$, and T_{res} represent the heat transfer coefficient of the refrigerant flow, the total heat transfer area of the tube, the refrigerant temperature at the center of the tube, and the thermal reservoir’s temperature, respectively. The heat transfer coefficient for the single-phase refrigerant flow is given by:

$$\lambda_{r.1} = \left(\frac{k_{1.cen} Nu_1}{D_h} \right), \quad (19)$$

where $k_{1.cen}$ and Nu_1 represent the thermal conductivity of the refrigerant and the Nusselt number of the single-phase refrigerant flow, respectively. The heat transfer coefficient for the two-phase refrigerant flow can be calculated as:

$$\lambda_{r.2} = \left(\frac{k_{cen.L} Nu_2}{D_h} \right), \quad (20)$$

where $k_{cen.L}$ and Nu_2 represent the thermal conductivity of the saturated liquid state of the refrigerant and the Nusselt number of the two-phase refrigerant flow, respectively.

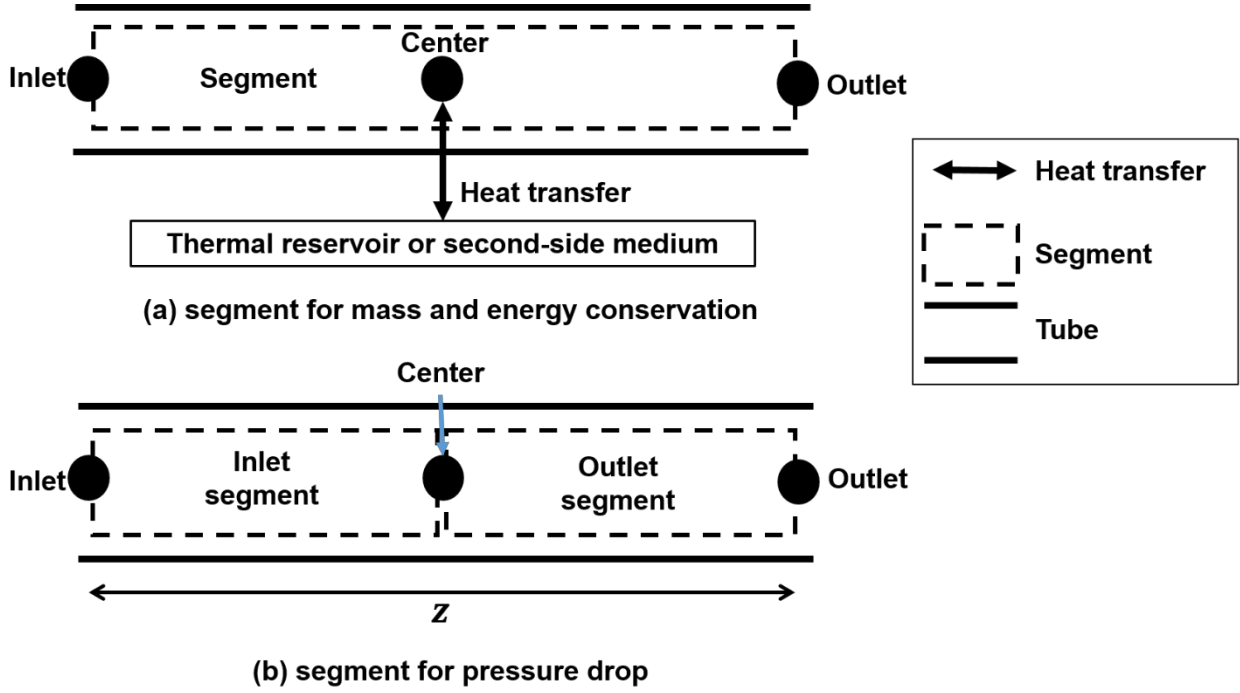


Fig. 6: Segment for calculation.

The pressure drop equation of refrigerant flow inside a tube can be written as follows:

$$\begin{aligned} \left(-\frac{dP}{dz}\right)_{total} &= \left(-\frac{dP}{dz}\right)_{friction} + \\ &\left(-\frac{dP}{dz}\right)_{acceleration} + \left(-\frac{dP}{dz}\right)_{gravitation} + \\ &\left(-\frac{dP}{dz}\right)_{minor}, \end{aligned} \quad (21)$$

In the right hand of the Eq. (21), the last term represents the pressure drop due to the change of the cross-section area of the refrigerant flow at the inlet and the outlet. Based on the above assumptions 2 and 3, the Eq. (21) is given by the following simplified form:

$$\left(-\frac{dP}{dz}\right)_{total} = \left(-\frac{dP}{dz}\right)_{friction} + \left(-\frac{dP}{dz}\right)_{acceleration}. \quad (22)$$

Fig. 6 (b) shows the segments for pressure drop

calculation. A single tube consists of two segments; the inlet segment and the outlet segment. The Eq. (22) is applied to each segment with the following form of pressure drop equations, respectively:

Inlet segment:

$$P_i - P_{cen} = \left(\frac{f G_i^2 v_{cen}}{2 D_h}\right) \left(\frac{z}{2}\right) + G_i |G_i (v_{cen} - v_i)|, \quad (23)$$

Outlet segment:

$$P_o - P_{cen} = \left(\frac{f G_o^2 v_{cen}}{2 D_h}\right) \left(\frac{z}{2}\right) + G_o |G_o (v_{cen} - v_o)|, \quad (24)$$

where z , f , and D_h represent the length of the tube, the Darcy friction factor, and the hydraulic diameter of the tube, respectively. Except for the Darcy friction factor and the Nusselt number, the mentioned equations (Eq. 16 - 24) consist of geometric parameters and thermodynamic properties of the refrigerant. The model and correlations of the Darcy friction factor and the Nusselt number which applied to the 'Pipe (2P)' and 'EvapPipe' are tabulated in Table 5.

Table 5. Model and correlations for friction factor and Nusselt number.

Friction factor	
Single-phase laminar flow	$f = \left(\frac{64}{Re_1}\right)^{.28}$
Single-phase turbulent flow	Haaland ²⁵⁾ $f = \left\{ -1.8 \log_{10} \left[\frac{6.9}{Re_1} + \left(\frac{\varepsilon/D_h}{3.7} \right)^{1.11} \right] \right\}^{-2}$
Two-phase flow boiling	Haaland ²⁵⁾ , using two-phase Reynolds number ($Re_2 = \frac{G_2 D_h}{\mu_2}$).
Nusselt number	
Single-phase laminar flow	$Nu_1 = 3.66$. ²⁸⁾
Single-phase turbulent flow	Gnielinski ²⁶⁾ $Nu_1 = \frac{\left(\frac{f}{8}\right)(Re_1 - 1000)Pr_1}{1 + 12.7\left(\frac{f}{8}\right)^{0.5}\left(Pr_1^{\frac{2}{3}} - 1\right)}$ Here, the friction factor is calculated by Haaland ²⁵⁾ .
Two-phase flow boiling	Fang ²⁷⁾ $Nu_2 = \frac{0.00061(C_1 + C_2)(Re_L Fa^{0.11})Pr_L^{0.4}}{\ln\left(\frac{1.024\mu_L}{\mu_{L,wall}}\right)},$ $C_1 = 41000Bo^{1.13} - 0.275,$ $C_2 = \left(\frac{x}{1-x}\right)^a \left(\frac{\rho_L}{\rho_V}\right)^{0.4},$ $a = \begin{cases} 0.48 + 0.00524(Re_L Fa^{0.11})^{0.85} - 5.9 \times 10^{-6}(Re_L Fa^{0.11})^{1.85} & \text{if, } (Re_L Fa^{0.11}) < 600 \\ 0.87 & \text{if, } 600 \leq (Re_L Fa^{0.11}) \leq 6000 \\ \frac{160.8}{(Re_L Fa^{0.11})^{0.6}} & \text{if, } 6000 < (Re_L Fa^{0.11}) \end{cases}$ $Re_L = \left(\frac{(1-x)GD_h}{\mu_L}\right),$ $Bo = \left(\frac{q}{Gh_{LV}}\right),$ $Fa = \left(\frac{(\rho_L - \rho_V)\sigma}{G^2 D_h}\right).$

In the present model, the Nusselt number of R-744 transcritical flow at the gas cooler is calculated by Gnielinski²⁶⁾ correlation. Yin et al.²⁹⁾ adopted the Gnielinski²⁶⁾ correlation for modeling R-744 transcritical flow in 0.79 mm of the refrigerant channel and reported that the calculated heat rejection rate is well-matched with their experimental heat rejection rate of the gas cooler, with an error of $\pm 2\%$. Rieberer³⁰⁾ compared six correlations for R-744 flow in 7.8 mm of smooth tubes. Among them, Gnielinski²⁶⁾ correlation showed the best prediction result that compared to the measured heat transfer coefficient within $\pm 30\%$ over a wide temperature range. Besides, according to Petterson et al.³¹⁾, Gnielinski²⁶⁾ correlation predicts well for cases of micro-scaled channels, also. In the present study, the hydraulic diameter of the refrigerant flow in the gas cooler is 2.8 mm; this value lies within the valid range mentioned in the previous literature^{29),30)} (0.79 mm to 7.8 mm).

Fang²⁷⁾ analyzed 2956 experimental data of R-744 flow boiling over a wide range of operating conditions ($-40\text{ }^{\circ}\text{C}$ to $27\text{ }^{\circ}\text{C}$ of evaporation temperature; $97\text{ kg m}^{-2}\text{s}^{-1}$ to $1400\text{ kg m}^{-2}\text{s}^{-1}$ of mass flux; 3.9 kW m^{-2} to 40 kW m^{-2} of heat flux) and hydraulic diameter of the flow channel (0.53 mm to 7.75 mm). Based on the database, Fang²⁷⁾ proposed a correlation of the Nusselt number for R-744 flow boiling by adding a non-dimensional number which takes into account the effect of bubble formation and departure. The proposed correlation predicts the boiling heat transfer coefficient of R-744 within $\pm 30\%$ error against 90% of the database. The operating conditions and the hydraulic diameter of the evaporator in the present manuscript lie in the applicable range of the Fang correlation. Thus, the Fang correlation is used for the Nusselt number calculation of R-744 flow boiling in the evaporator.

2.3.3 Orifice

An orifice as an expansion device was constructed employing the 'Variable Local Restriction (2P)' model in Fig. 3 (a). The model operates under the following assumptions:

1. The orifice is a rigid body.
2. Heat gain or loss of the orifice body is neglected.
3. The expansion process of the refrigerant flow is isenthalpic.
4. The refrigerant flow is one dimensional.

Mass and energy conservation equations are as follows:

$$\dot{m}_{ex,i} = \dot{m}_{ex,o}, \quad (25)$$

$$h_{ex,i} = h_{ex,o}. \quad (26)$$

Here, the subscript, *ex*, represents the expansion device. The refrigerant mass flow rate due to the irreversible pressure drop is calculated as follows:

$$\dot{m}_r = C_D A_{c,ori} \left[\frac{2(P_i - P_o)}{v_{ori} K} \right]^{0.5}, \quad (27)$$

where C_D , $A_{c,ori}$, v_{ori} , and K represent the flow coefficient, the cross-section area of the orifice, the specific volume of the refrigerant (defined at the orifice), and the correction factor, respectively. The correction factor (K) is applied by the Simscape's default setting to avoid encountering numerical errors and implement a smooth computing procedure during simulations. The flow coefficient (C_D) of a certain valve is specified based on its geometric shape and size; in the present model, the flow coefficient was set the fixed default value, 0.64.

2.4 Simulation procedure

Fig. 7 describes a simulation flow chart. When a transient simulation starts, Simscape solver builds up a physical network of the heat pump system model based on the setting parameters in each component model and the input values (Phase A). Then, a system of equations is constructed based on the constructed physical network (Phase B). The solver analyzes the constructed system of equations and eliminates variables that are not needed to solve the system of equations. Then, the solver computes the initial conditions by finding initial values (at $t=0$) for all variables in each component model (Phase C). These initial values must be exactly satisfied with the whole physical conservation equations. When the solver finds the initial values, it starts computing by the end of the set simulation time (Phase D). After each simulation finished, the result values such as the refrigerant mass flow rate, thermodynamic states at each point, capacities of each HX, and compressor's power, were obtained in each time-step (Phase E).

In this study, one of the variable time-step implicit solvers, 'ode23t', was selected as the solver of simulation the heat pump system. It is because the 'ode23t' showed the shortest running time compared to among other variable time-step implicit solvers.

For the steady-state simulations, a long enough time, 3600 seconds, was set for the transient simulation time. After each transient simulation finished, the operating parameters of the heat pump system model, such as the refrigerant mass flow rate, pressure, and temperature were checked. When these values lie within the experimental uncertainties without any variation, it was decided that the heat pump system model reached a steady-state.

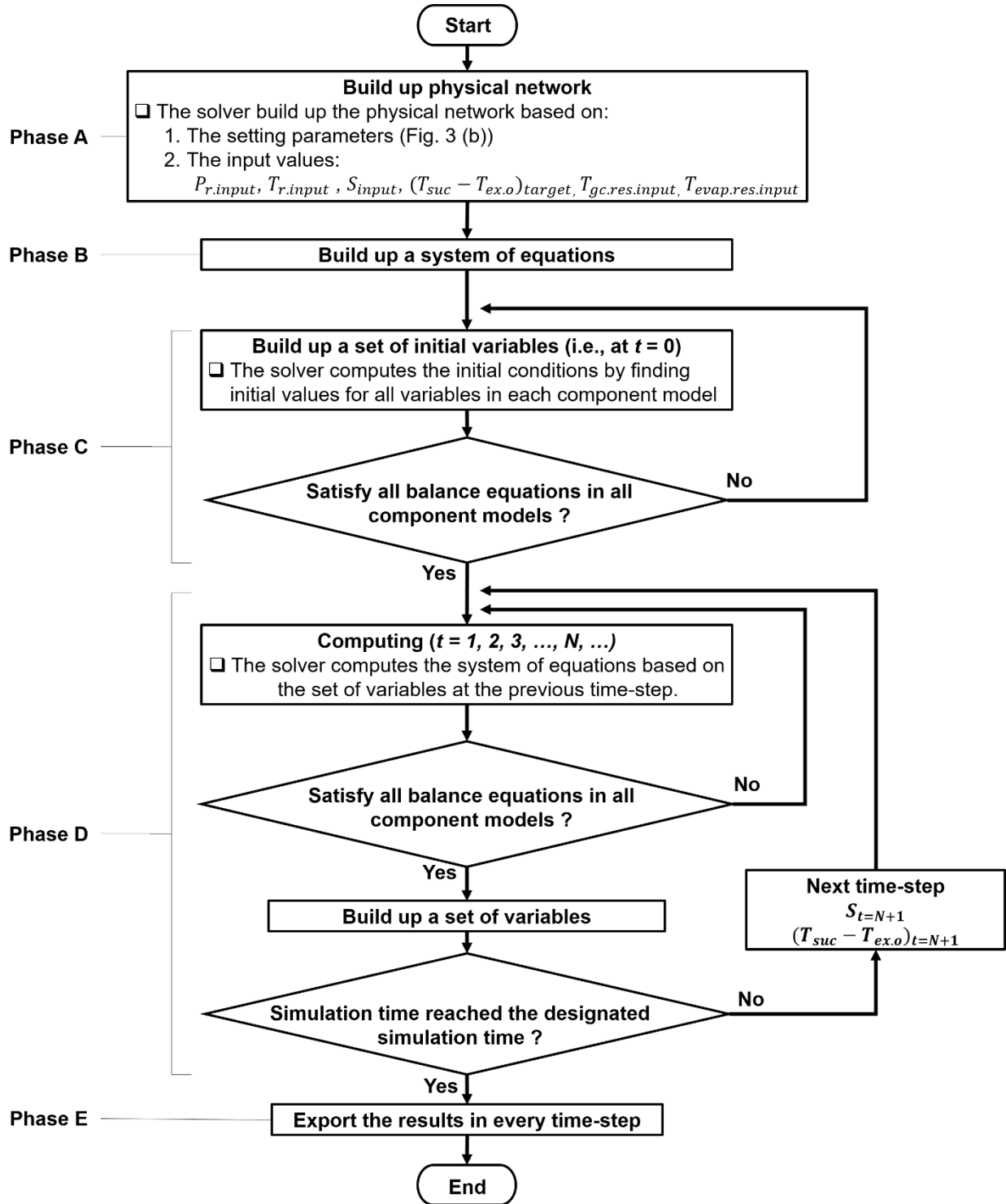


Fig. 7: Flow chart.

3. Results and discussion

3.1 Steady-state performance

A set of experimental conditions in steady-states is summarized in Table 6. Each experiment was carried out to achieve the target cooling and heating capacities, primarily. The adjusting parameters for the cooling capacity were 1.6 and 1.8 kW for the evaporator. The heating capacity of the gas cooler changes from 2.4 to 3 kW with a step of 0.2 kW. The experimental results corresponding to each case are tabulated in Table 7.

Table 8 shows simulation conditions corresponding to each experimental case. The rotation speed of the compressor model was selected as the identical value with the rotation speed obtained from each experiment. The target temperature difference between the suction and the expansion device outlet, $(T_{suc}-T_{ex.o})_{target}$, was adjusted to fulfill the experimental mass flow rate of the refrigerant. The simulation results are tabulated in Table 9.

Fig. 8 presents the prediction error of the calculated refrigerant mass flow rate, capacities of each HX, and

COPs. The simulation results show a good agreement for the refrigerant mass flow rate, heating, and cooling capacity of each HX, compressor power, and COPs. As shown in Fig. 8 (a), the model predicts the measured refrigerant mass flow rate in a maximum error of $\pm 6\%$. Also, the model predicts both the cooling capacity of the evaporator and the heating capacity of the gas cooler within an error of $\pm 5\%$. In Fig. 8 (b), the simulation model under-predicts the compressor power. The compressor model predicts similar discharge pressures with less input power as compared to the corresponding actual compressor. As the suction pressure in an actual case has the same value as that of its ideal case; the system needs higher compressor power to lift the refrigerant pressure to the same level as in the ideal case. It is because of the irreversible losses during the actual compression process. In the present system model, the compressor is simply modeled with several assumptions, thus, it is hard to consider all the sources of losses in an actual compression process. Consequently, as shown in Fig. 8 (c), the simulation model over-predicts the COP values of the experimental system within an error of 7%.

Table 6. Experimental conditions in steady-state.

Case	Target	Gas cooler				Evaporator				Comp.	EEV
		Water		Refrigerant		Water		Refrigerant			
		$T_{w.i.gc}$	$\dot{V}_{w.gc}$	$P_{r.i.gc}$	$T_{r.i.gc}$	$T_{w.i.evap}$	$\dot{V}_{w.evap}$	$P_{r.i.evap}$	$h_{r.i.evap}$		
	kW	°C	L/h	MPa	°C	°C	L/h	MPa	kJ/kg	rev/s	-
1	$\dot{Q}_{evap} = 1.6$	29.9	126.6	8.29	75.0	20.4	90.8	3.92	295.8	72	225
2	$\dot{Q}_{evap} = 1.8$	30.0	137.3	8.32	76.6	20.4	100.0	3.94	294.4	78	237
3	$\dot{Q}_{evap} = 1.6$	29.8	118.4	8.36	68.4	20.4	134.9	4.45	295.0	60	219
4	$\dot{Q}_{evap} = 1.8$	29.8	128.5	8.43	70.0	20.4	147.3	4.44	286.7	62	179
5	$\dot{Q}_{evap} = 1.8$	29.8	137.2	10.0	86.0	20.4	152.9	4.45	270.9	60	127
6	$\dot{Q}_{gc} = 2.4$	20.1	83.8	7.83	65.4	15.6	253.6	4.32	283.8	68	245
7	$\dot{Q}_{gc} = 2.6$	20.1	91.1	7.85	66.3	15.6	273.7	4.32	283.1	74	263
8	$\dot{Q}_{gc} = 2.6$	20.1	89.3	7.92	66.4	15.6	270.9	4.32	272.2	68	176
9	$\dot{Q}_{gc} = 2.8$	20.1	97.2	7.95	66.8	15.7	289.4	4.31	271.9	74	192
10	$\dot{Q}_{gc} = 3.0$	20.1	103.7	8.29	69.1	15.6	318.7	4.30	247.9	70	142
11	$\dot{Q}_{gc} = 3.0$	24.9	67.6	10.0	80.5	20.5	310.7	4.85	269.2	68	159
12	$\dot{Q}_{gc} = 2.8$	24.9	64.2	10.1	80.3	20.5	296.8	4.87	266.6	64	146
13	$\dot{Q}_{gc} = 3.0$	24.9	69.0	10.1	82.2	20.5	318.0	4.85	263.9	68	150
14	$\dot{Q}_{gc} = 2.8$	24.9	65.1	10.4	83.8	20.5	302.7	4.85	256.9	62	126
15	$\dot{Q}_{ac} = 3.0$	24.9	67.8	10.5	85.0	20.5	315.7	4.85	256.3	64	129

Table 7. Experimental results in steady-state.

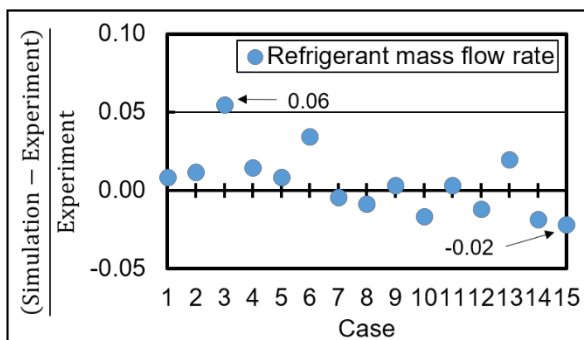
#	Gas cooler				Evaporator					ΔT_{sup}	\dot{m}_r	\dot{W}	COP_c	COP_h
	Water		Refrigerant		Water		Refrigerant							
	$T_{w.o.gc}$	$\dot{Q}_{w.gc}$	$T_{r.o.gc}$	$\dot{Q}_{r.gc}$	$T_{w.o.evap}$	$\dot{Q}_{w.evap}$	$T_{r.o.evap}$	$P_{r.o.evap}$	$\dot{Q}_{r.evap}$					
	°C	kW	°C	kW	°C	kW	°C	MPa	kW					
1	45.0	2.21	32.5	2.31	4.9	1.63	5.4	3.77	1.73	2.5	45.4	0.63	2.59	3.51
2	45.1	2.41	32.3	2.52	5.0	1.78	5.7	3.76	1.88	2.9	48.5	0.69	2.58	3.49
3	45.0	2.07	32.5	2.16	9.9	1.63	11.9	4.32	1.72	3.5	45.2	0.49	3.33	4.22
4	45.0	2.26	31.3	2.35	9.9	1.79	12.1	4.44	1.88	3.8	46.3	0.52	3.44	4.35
5	45.1	2.42	29.5	2.54	10.0	1.85	11.9	4.33	1.95	3.5	43.6	0.65	2.85	3.72
6	45.0	2.42	29.3	2.52	9.0	1.94	10.7	4.16	2.04	3.8	48.9	0.52	3.73	4.65
7	45.0	2.63	29.2	2.73	9.0	2.10	10.3	4.14	2.20	3.6	52.5	0.57	3.68	4.61
8	45.0	2.58	27.0	2.68	9.0	2.08	10.6	4.16	2.19	3.8	48.7	0.52	4.00	4.96
9	45.0	2.90	27.0	2.90	9.1	2.24	8.7	4.12	2.35	3.1	52.6	0.58	3.86	5.00
10	45.0	3.00	20.4	3.12	9.0	2.44	9.1	4.14	2.56	2.4	50.2	0.58	4.21	5.17
11	63.0	2.99	29.2	3.11	13.9	2.36	14.6	4.68	2.46	2.8	55.4	0.71	3.32	4.21
12	63.0	2.83	28.5	2.96	13.9	2.25	14.3	4.68	2.35	2.6	52.4	0.67	3.36	4.22
13	62.9	3.04	27.6	3.16	14.0	2.41	14.7	4.64	2.51	3.2	54.4	0.72	3.35	4.22
14	63.1	2.88	25.6	3.01	13.9	2.30	14.5	4.68	2.40	2.7	50.3	0.69	3.33	4.17
15	62.9	2.99	25.5	3.13	14.0	2.38	14.6	4.68	2.48	2.0	51.8	0.72	3.31	4.15

Table 8. Simulation conditions in steady-state.

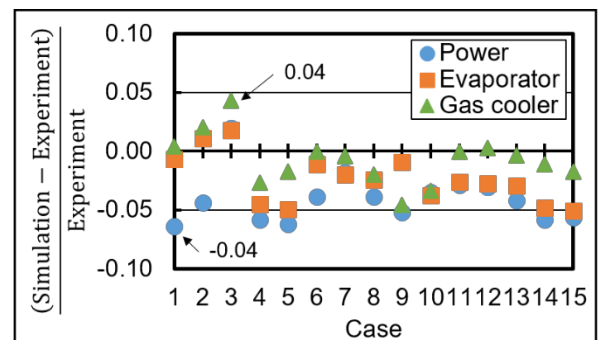
Case	Input value							
	Gas cooler			Evaporator			Expansion device	Comp.
	$T_{gc.res.input}$	$P_{r.gc.input}$	$T_{r.gc.input}$	$T_{evap.res.input}$	$P_{r.evap.input}$	$T_{r.evap.input}$	$(T_{suc} - T_{ex.o})_{target}$	S_{input}
	°C	MPa	°C	°C	MPa	°C	°C	rev/s
1	32.0	8.3	32.0	4.5	4.5	11.0	4.0	72
2	32.0	11.0	32.0	4.5	4.5	11.0	4.0	78
3	31.0	7.8	32.0	9.5	4.5	11.0	2.0	60
4	31.1	7.9	32.0	9.5	4.5	11.0	3.5	62
5	29.2	8.3	32.0	9.5	4.5	11.0	4.0	60
6	29.0	8.3	32.0	8.3	4.5	11.0	2.5	68
7	28.5	8.2	32.0	8.3	4.5	11.0	4.0	74
8	26.0	8.9	32.0	8.3	4.5	11.0	4.0	68
9	26.0	8.9	32.0	8.3	4.5	11.0	3.5	74
10	20.0	9.0	26.0	8.0	4.5	11.0	3.5	70
11	28.0	10.0	30.0	13.0	4.8	13.0	3.5	68
12	27.0	10.5	30.0	13.0	4.8	13.0	4.0	64
13	27.0	10.4	30.0	13.0	4.8	13.0	3.5	68
14	24.0	12.3	30.0	12.5	4.8	13.0	4.0	62
15	24.0	12.3	30.0	12.5	4.8	13.0	4.0	64

Table 9. Simulation results in steady-state.

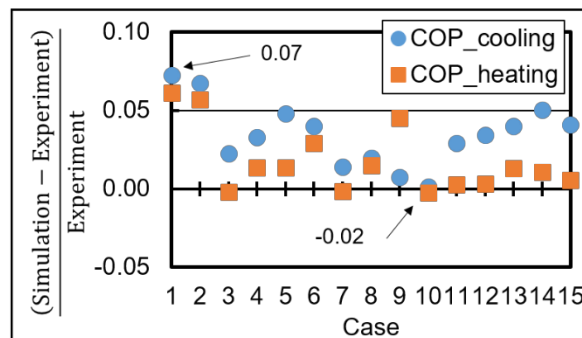
Case	Result						
	\dot{m}_r	ΔT_{sup}	\dot{Q}_{evap}	\dot{Q}_{gc}	\dot{W}	COP_c	COP_h
	kg/h	°C	kW	kW	kW	-	-
1	45.8	4.9	1.62	2.22	0.59	2.75	3.76
2	49.1	4.9	1.80	2.46	0.66	2.73	3.73
3	47.7	2.9	1.66	2.16	0.50	3.32	4.32
4	47.0	4.9	1.71	2.20	0.49	3.49	4.49
5	44.0	4.0	1.76	2.38	0.61	2.89	3.90
6	50.6	3.5	1.92	2.42	0.50	3.84	4.84
7	52.3	5.7	2.06	2.62	0.56	3.68	4.68
8	48.3	4.8	2.03	2.53	0.50	4.06	5.06
9	52.8	4.4	2.22	2.77	0.55	4.04	5.04
10	49.4	3.5	2.35	2.90	0.56	4.20	5.18
11	55.6	3.6	2.30	2.99	0.69	3.33	4.33
12	51.8	4.0	2.19	2.84	0.65	3.37	4.37
13	55.5	3.5	2.34	3.03	0.69	3.39	4.39
14	49.4	4.0	2.19	2.85	0.65	3.37	4.38
15	50.7	4.1	2.26	2.94	0.68	3.32	4.32



(a) Mass flow rate



(b) Compressor power and capacity of HXs



(c) COP

Fig. 8: Comparison results; (a) refrigerant mass flow rate, (b) compressor power and capacity of each HX, (c) COPs.

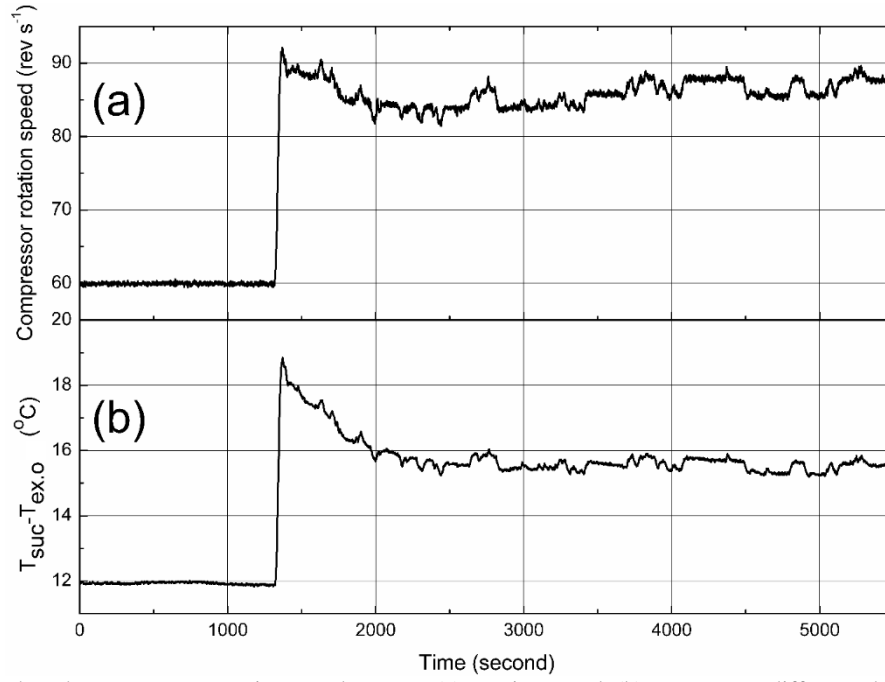


Fig. 9: Experiment when the compressor rotation speed step-up; (a) rotation speed, (b) temperature difference between at the suction and the EEV outlet.

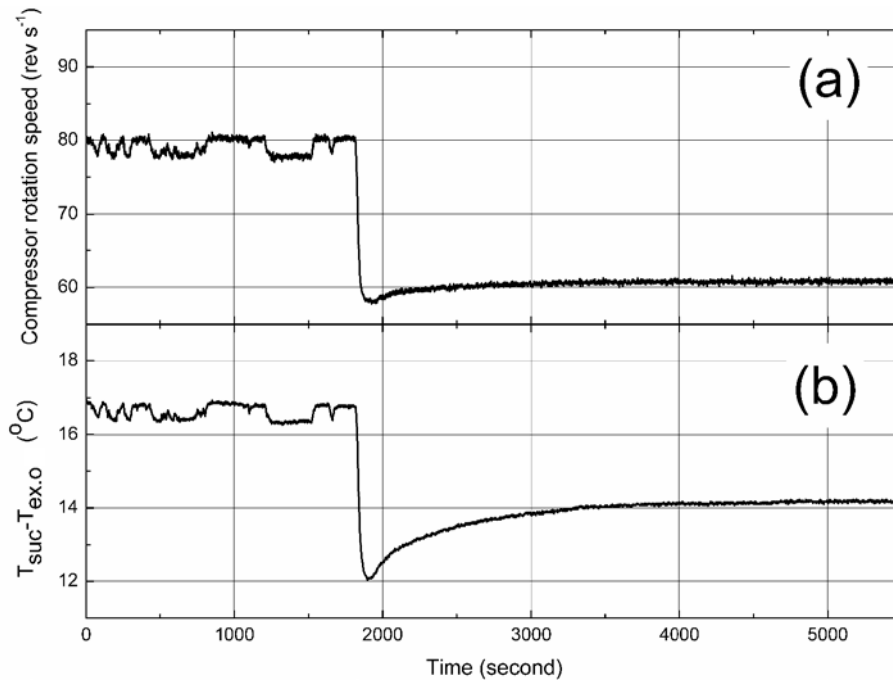


Fig. 10: Experiment when the compressor rotation speed step-down; (a) rotation speed, (b) temperature difference between at the suction and the EEV outlet.

3.2 Transient performance

In the case of transient conditions, experiments have been carried out with changing compressor rotation speed. Fig. 9 (a) depicts the experimental condition when the compressor rotation speed was suddenly increased from 60 rev s^{-1} to 87 rev s^{-1} . During the experiment, the feed pump at the cooling water loop supplied the cooling water to the gas cooler with 128.7 L/h of flow rate. The measured water temperature at the gas cooler inlet was

30°C . The heating water was supplied to the evaporator with 99.9 L/h of flow rate and the measured temperature was 20°C at the evaporator inlet. In each water loop, the flow rate of supplying water was kept constant and the measured water temperature at each HX inlet did not change during the experiment. Fig 9 (b) shows the measured temperature difference between the suction and the EEV outlet, $(T_{suc} - T_{ex.o})$, by the experimental EEV's opening. In the present simulation, the experimental

compressor speed (Fig. 9 (a)) was used for the input value of the compressor model. The measured temperature difference (Fig. 9 (b)) was also used for the input value to the 'Control box Cross-section area' in the expansion device model (Fig. 3. (a)), as the target value of the temperature difference, $(T_{suc}-T_{ex.o})_{target}$. The temperature of the thermal reservoir at the gas cooler model was set to 31 °C and the temperature of the thermal reservoir at the evaporator model was set to 8.3 °C. In the gas cooler, the input values of the refrigerant pressure and temperature were set to 7.7 MPa and 32 °C, respectively, while the input values of the refrigerant pressure and temperature were set to 4.5 MPa and 11 °C, respectively, in the evaporator.

Fig. 10 (a) shows the compressor rotation speed with time when the rotation speed was suddenly decreased from 80 rev s⁻¹ to 60 rev s⁻¹. In this experiment, the cooling water temperature at the gas cooler inlet was fixed to 30 °C and the heating water temperature at the evaporator inlet was fixed to 20 °C. The feed pump in the cooling water loop supplied cooling water to the gas cooler with 129.6 L/h of flow rate. In the heating water loop, the pump supplied heating water with 118.4 L/h of flow rate. Fig. 10 (b) shows the measured temperature difference between the suction and the EEV outlet, $(T_{suc}-T_{ex.o})$. As same as the case that the compressor speed suddenly increases, the measured compressor speed was used for the input value of the compressor model and the measured temperature difference was used for the input value for the target temperature difference, $(T_{suc}-T_{ex.o})_{target}$, in the expansion device model. The temperature of the thermal reservoir at the gas cooler model was set to 29 °C and the temperature of the thermal reservoir at the evaporator model was set to 8.3 °C. In the gas cooler, the input values of the refrigerant pressure and temperature were set 7.9 MPa and 32 °C, respectively. The refrigerant pressure of 4.5 MPa and the refrigerant temperature of 11 °C were input to the evaporator model.

The comparison results between the simulations and the experiments are described in Fig. 11 and Fig. 12. The results in Fig. 11 corresponds to the case when the

compressor speed suddenly increases and, the results in Fig. 12 corresponds to the case when the compressor speed suddenly decreases, respectively.

As shown in Fig. 11 and Fig. 12, the comparison results show that the simulation model is a good agreement with the dynamic behavior of the experimental system and the corresponding performance change. However, there is a significant difference between the simulation and the experimental results in the gas cooler pressure and the heating capacity of the gas cooler, right after the compressor speed changes; this difference is commonly observed in Fig. 11 and Fig. 12. The corresponding simulation's results show an immediate response against the change, however, the corresponding experimental results show a gradual response against the change. This difference is mainly due to the compressor model. The present compressor model does not consider heat transfer between the compressor body and the ambient, and mechanical losses. During the compression process in an actual compressor, the input power does not fully contribute to increasing the refrigerant enthalpy, i.e., it is not fully utilized to the refrigerant's compression process. There is a loss of the part of the input power, therefore, it takes a significant time to supply the refrigerant with enough enthalpy. As shown in Fig. 11 (c and e) and Fig. 12 (c and e), the simulation results show that the calculated suction pressure and the cooling capacity at the evaporator are a good agreement with the response of the experimental system even though right after the compressor speed changes. This is because when the compressor speed changes in an actual system, the operating parameters in the low-pressure side of the system, such as the suction pressure, the suction temperature, and the suction flow rate immediately response to this change; therefore, the cooling capacity of the evaporator also responses this immediate change in an actual case. On the other hand, when the system becomes stable after the compressor speed changes, the simulation model recovers the prediction error within $\pm 7\%$.

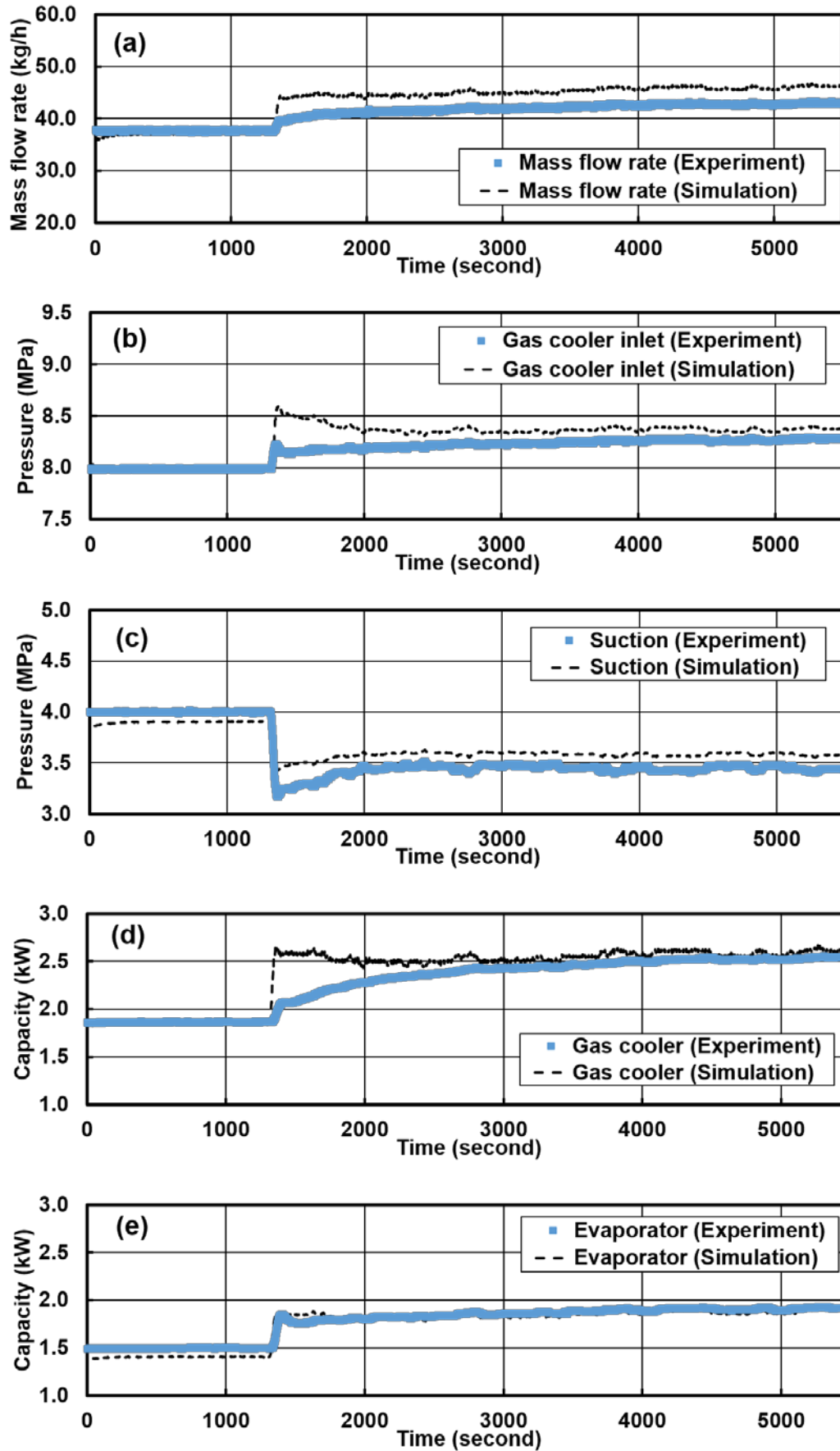


Fig. 11: Comparison results when the compressor rotation speed step-up; (a) refrigerant mass flow rate, (b) gas cooler inlet pressure, (c) suction pressure, (d) heating capacity, (e) cooling capacity.

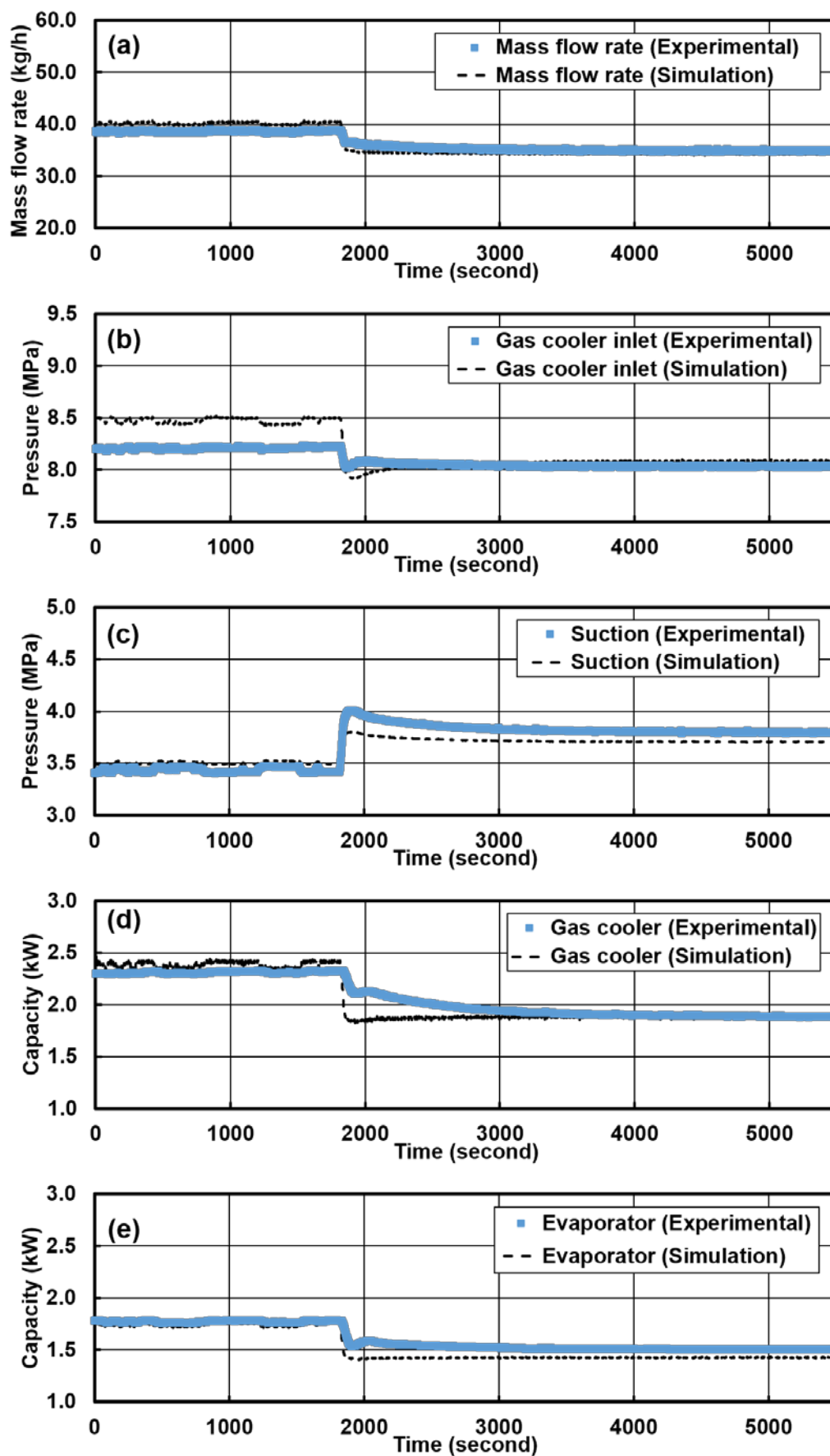


Fig. 12: Comparison results when the compressor rotation speed step-down; (a) refrigerant mass flow rate, (b) gas cooler inlet pressure, (c) suction pressure, (d) heating capacity, (e) cooling capacity.

4. Conclusion and limitations

To develop smart-controlled thermal energy systems operated by real-time demand, it is key to understand the dynamic behaviors of the comprising devices; for that, pre-validated dynamic models are necessary. In the present study, a dynamic model for an R-744 transcritical heat pump system was developed utilizing the Simscape library. Using the model, simulations were conducted for both transient and steady-state conditions. Experiments were also conducted using R-744 as a working fluid to validate the simulation model. The simulation results, such as refrigerant mass flow rate, cooling, and heating capacity, compressor power, and COPs were compared to that of the experimental data. The comparison results demonstrate that the simulation model can predict the performance of the actual system within $\pm 7\%$ error in given steady-state conditions as well as the dynamic behavior of the actual system. However, a significant deviation from the experimental results was observed in the simulation results, when the compressor rotation speed changes. This is because the present compressor model does not take into account the effects of heat transfer with the ambient, and mechanical losses. For the improvement of the model, detailed compressor modeling will be required.

On the other hand, it is known that the heat transfer characteristic of R-744 abruptly changes in the transcritical region, since the thermophysical properties of R-744 changes dramatically near the pseudo-critical line. In the present gas cooler model established at the Simscape platform, the unit segment of the R-744 flow was set to the same size as the single tube of the experimental gas cooler. According to the validation results, the model shows decent prediction results in system-scaled performance and behavior, however, considering the set segment size, it will be difficult to follow up with the sudden change of the refrigerant heat transfer characteristic. This limitation may be solved by decreasing the segment size while this can cause an additional computational cost due to the increment in the number of segments. For improving the model, it is necessary that the size of the unit segment in gas cooler tubes are adjusted based on the sensitivity test of the segment size.

Acknowledgments

Not applicable.

Nomenclature

A	heat transfer area (m ²)
A_c	cross-section area (m ²)
Bo	boiling number (-)
C_D	flow coefficient (-)
C_P	specific heat at constant pressure

COP	(kJ kg ⁻¹ K ⁻¹)
D_h	coefficient of performance (-)
f	hydraulic diameter (m)
G	Darcy friction factor (-)
h	mass flux (kg m ⁻² s ⁻¹)
h_{LV}	specific enthalpy (kJ kg ⁻¹)
k	specific latent heat (kJ kg ⁻¹)
\dot{m}	thermal conductivity (W m ⁻¹ K ⁻¹)
Nu	mass flow rate (kg s ⁻¹)
P	Nusselt number (-)
Pr	pressure (MPa)
\dot{Q}	Prandtl number (-)
q	heat transfer rate (W)
Re	heat flux (W m ⁻²)
S	Reynolds number (-)
T	compressor rotation speed (rev s ⁻¹)
ΔT_{sup}	temperature (K or °C)
	degree of suction superheating (= $T_{suc} - T_{sat.@P_{suc}}$) (K or °C)
t	time (second)
U	uncertainty (-)
u	specific internal energy (kJ kg ⁻¹)
V	volume (m ³)
V_{comp}	compressor swept volume (m ³)
\dot{V}	volumetric flow rate (m ³ s ⁻¹)
v	specific volume (m ³ kg ⁻¹)
\dot{W}	compressor power (W)
x	vapor quality (-)
z	length (m)

Greek symbols

Δ	difference
ε	absolute roughness (m)
η	efficiency (-)
ζ	property
λ	heat transfer coefficient (W m ⁻² K ⁻¹)
μ	dynamic viscosity (Pa s)
ρ	density (kg m ³)
σ	surface tension (N m ⁻¹)
Ψ	reduced data
ψ	measured data

Subscripts

1	singe-phase
2	two-phase
c	cooling
cen	center
dis	discharge
evap	evaporator
ex	expansion device
exp	experimental
gc	gas cooler
i	inlet
input	input
L	saturated liquid state
o	outlet

ori	orifice
pred	prediction
r	refrigerant
s	isentropic
res	thermal reservoir
suc	suction
sup	superheating
V	saturated vapor state
v	volumetric
w	water
wall	tube wall

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