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On the nozzle geometry of a transferred plasma cutting torch for the arc stabilization with swirl flow

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In the plasma cutting process, the arcjet with high specific energy is required to get good cut quality in the dross-free cutting. For this purpose, it is necessary to stabilize the arc with gas flow rate lower For this purpose, it is necessary to stabilize the arc with gas flow rate lower than the conventional torches. The present paper discusses the nozzle shape that effectively stabilizes the arc by the swirl flow. The pressure measurement inside the large-scale model-nozzle indicates that the nozzle shape can be optimized to create a large pressure gradient for the arc stabilization with a constant gas flow rate. The metal cutting test by the nozzle with optimum shape shows a remarkable The metal cutting test by the nozzle with optimum shape shows a remarkable improvement in arc stabilization.

1. Introduction

 The arc stabilization by the swirl flow has been used in the various types of plasma torches¹⁾. The arc stabilization in a torch is very important for the stable operation of the plasma cutter, However, the characteristics of the swirl flow inside a plasma torch has not been well understood, and the geometry of the torch has not been optimized for the effective generation of the swirl flow. If the arc is stabilized by a swirl flow with a small flow rate, it is possible to increase the specific energy of the transferred arcjet. This contributes to the cut quality in plasma cutting process, especially to the dross-free cutting²⁾.

To realize the dross-free cutting, it is necessary²⁾ to develop a new torch that stabilizes the arc with the flow rate lower than the critical value where the flow is considered laminar. To stabilize an arc effectively, it is required to optimize the geometry of the discharge space, where the arc is sustained with the high pressure gas wall created by the swirl flow. The centrifugal force of the swirl flow creates large pressure gradient in the discharge space, and this force takes a minimum value at the center of the torch and a maximum value on the surface of the nozzle wall. An arc column is stably sustained in the low pressure region near the center of a torch. The parameter that expresses the effect of a swirl flow is static pressures on the wall surfaces of the nozzle and electrode which constitute the discharge space in the torch. By measuring these static pressures, the pressure distribution in the swirl flow is estimated. When the pressure gradient is large, the arc becomes stable, so that the conventional torches need large flow rate to increase the centrifugal force for the stable arcing. This results in the arcjet having low specific energy and degrades the cut quality.

 The structure of the swirl flow depends mainly on the nozzle shape, so that it is very important to determine its shape to create the maximum pressure gradient under the

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Using a large-scale torch model made of acrylic-resin, the condition of constant flow rate. pressure distribution in the discharge space was measured for various nozzle models with The metal cutting test was also performed to compare the performance different shapes. of the nozzle with the optimum shape with that of the commercial nozzle of Komatsu G940 torch.

The nozzle with a mixing chamber 2.

Figure 1 shows the schematic diagram of the nozzle and cathode, where d is the gap distance between the nozzle wall and the cathode surface, ν the exit velocity of the arciet. The center of the nozzle does not coincide accurately with that of the electrode due to the manufacturing and setting errors. Even the minute defference caused by these errors creates the difference in the gap distance and results in the differences of the velocity and Consequently, the arciet is deflected according to the pressure across the nozzle exit. magnitude of the difference in concentricity, as it leaves the nozzle. The jet with large deflection causes a cut surface to be inclined and results in unacceptable precision of the cut part³.

jet deviation and the inclined cut surface.

To solve this serious problem, a mixing chamber⁴ was devised to equalize the flow pressure before the plasma leaves the nozzle, as shown in Fig. 2. This mixing chamber showed a remarkable effect to prevent the jet deviation⁴⁾. This effect is explained in detail in the latter part of this paper.

Figure 3 shows the cross sectional view of the torch head (Komatsu G940). In this Figure, ΔL is the distance between the top surface of the electrode and the upper surface of the mixing chamber, D the diameter of the mixing chamber, d_1 the entrance length, d_2 the depth of the mixing chamber and d_3 the length of the acceleration being fixed to 3.4 mm in The swirler has four equally spaced injection orifices tangential to a this experiment.

Mixina

chamber

Insulator

[A-A cross section of the swirler]

[Scales to be optimized]

Torch structure and scale parameters to be Fig. 3 optimized.

cylindrical surface and perpendicular to the center axis of the torch

3. Pressure measurement in the torch with the large-scale model

Figures 4 and 5 show the schematic diagrams of the large-scale torch model and the measurement system, respectively. In Fig. 4, p_e is the static pressure at the center of the top surface of the electrode, p_{vr} the static pressure at the bottom of the wall of the mixing chamber and p_n the static pressures measured by pressure holes equally spaced with an angle of 90° at the the orifice middle α f passage. The

Fig. 4 The large-scale torch model and the measured pressures.

Fig. 5 Pressure measurement system.

maximum difference among the four values of p_n is denoted by dp_n .

 The large-scale model is five times larger than the commercial torch (Komatsu G940) . Each component of the model can be replaced to test the different geometries for the change in experimental conditions. The electrode position relative to the nozzle is adjustable in horizontal and vertical direction to simulate the manufacturing and setting errors. The distance between the center of the nozzle and electode is denoted by Δr .

 The gas flow rate necessary for this model experiment was estimated by considering that the Reynolds number of the flow inside the model was equal to that of the actual torch.

To measure the static pressures at each point in the model, small holes with 0.3 mm in diameter are drilled on the walls of the nozzle and electrode as shown in Fig. 4. These are connected to the pressure transducers for automated data acquisition with a laboratory computer as shown in Fig. 5.

 The location of the pressure holes is determined by the following two considerations. Firstly, the large centrifugal force of the swirl flow makes the p_e small and the p_{vr} large. Secondly, the dp_n is a measure of the nonuniformity of the pressure distribution on the orifice exit plane; when the value of $d\rho_n$ is zero, the arcjet can be regarded as straight and no flow deviation occurs.

4. Results and discussion

 The experimental results obtained with the large-scale torch model by the present pressure measurement system are described in this chapter. All the symbols used in the following figures are consistent with the notations in Figs. 3 and 4.

4.1 Effect of the mixing chamber

To clarify the effect of the mixing chamber experimentally, the values of dp_n are

measured for two different nozzles. One has a mixing chamber with d_1 =11.8 mm, d_2 $=3.0$ mm, $D=20.0$ mm and $\Delta L=0.0$ mm and the other a simple conically convergent nozzle with the same length and exit diameter. Figure 6 shows the dp_n as a function of the distance Δr between the center of the nozzle and electrode, where m is the mass flow rate of the working gas. It is clear from this figure that the mixing chamber has a distinct effect for equalizing the pressure difference.

Fig. 6 Effect of the mixing chamber on dp_n as a function of Δr .

4.2 Effect of the displacement ΔL between nozzle and electrod

To clarify the effect of the relative position of the nozzle and the electrode, the ef- ΔL on the pressure distribution was investigated for $d_2\!=\!1.5$ and 3.0 mm. Figure 7 to (c) shows the plots of p_e , p_{vr} and dp_n as a function of ΔL . This figure indicate for $d_2=3.0$ mm, the best condition $(d p_n$ minimum) is obtained at $\Delta L\!=\!0;$ (ii) for d_2

function of ΔL . (a) p_e vs. ΔL (b) p_{vr} vs. ΔL (c) dp_n vs. ΔL

mm, almost no effect is seen to decrease the balue of flow condition can be obtained when ΔL is zero and d_2 is greater than or equal to 3. This indicates that the desirable

4.3 Effect of the mixing chamber depth d

To determine the optimum value of d_2 , the pressures were measured by **Figure 8 (a)** to (c) shows the measured pressures p_e , p_vr and dp_n as a fu This Clearly, the pressure difference dp_n for $d_2\text{=}1.5$ mm differs from the other two means that the mixing chamber does not work properly when the d_2 is excess Although there is no remarkable difference between the data for $d_2 = 3.0$ and 6 p_{vr} is small for the large d_2 . This tendency may generate an unstable arc such as caused by kink instability.

Fig. 9 Pressure profiles for different Δr as a function of D . (a) p_e vs. D (b) p_{vr} vs. D (c) dp_n vs. D

4.4 Effect of the mixing chamber diameter D

Figure 9 (a) to (c) shows the p_e , p_{vr} and dp_n as a function of the diameter of the mixing chamber for $\Delta r = 0.0$ and 0.5 mm. In this figure, the dp_n has a minimum value at $D = 20$ These results strongly suggest that there exists an optimum value of D. mm.

4.5 Effect of the entrance length d_1

Figure 10 (a) and (b) shows the effect of d_1 on the measured pressures. The p_e and p_{vr} are found to change linearly with d_1 . The large d_1 decreases the centrifugal force of the swirl flow by its large frictional loss. This figure also shows that there exists a minimum value of dp_n at $d_1 = 12$ mm.

5. The cutting performance with the optimized nozzle

To examine the validity of the experimental results obtained with the Iarge-scale torch

model and the pressure measurement system, the nozzle was made that had a similar size with the plasma cutter and was optimized according to the results of the present experiments. This nozzle was tested in the actual cutting process. The double arcing was caused directly by the arc instability inside the torch and could be used as a measure of the cutter performance. Therefore, the total number of the double arcing occurred during the cutting process was selected as a parameter for the evaluation. The test was performed by cutting the carbon-steel• plate with the thickness of 1.6 mm under the condition of dross-free cutting²⁾. Figure 11 and Table 1 show the test results and the nozzle specifications used in this test, respectively. In the figure, ϕ is the exit diameter of the nozzle, I the arc current and p_0 the supply pressure of the working gas. In the table, the optimized nozzle is denoted by "Proto" and the conventional nozzle by "G940". The same test procedure was repeated three times for these two nozzles. These results confirm the performance of the optimized nozzle superior to that of the conventional one.

6. Conclusions

 The optimum nozzle geometry for a plasma cutting torch was studied by using a large-scale torch model. The pressure measurement inside this model gives the

Cathode Irregular discharge Main arc due to arc instability ,
Nozzle Stand off \mathcal{S} is a set of \mathcal{S} distance Work plate {The double arcing] of double arcing 250 i, \sim Proto (N=1)
 $\left| \begin{array}{c} -\Delta - \text{ Proto (N=2)} \\ \text{Noting gas} \end{array} \right|$ 200 \bigoplus G940 (N=1)
-A-G940 (N=2) l"' ll. $\frac{1}{\sqrt{2}}$ = G940 (N=3)
 ϕ =0.6 mm, I=27 A 150 \cdot n Early
Early III $Stand$ off $= 2.0$ mm 1OO $p_0 = 3.5 \times 10^5$ Pa nai
Tai 50 t------- i i la anno i o-- -88 - -"+ ---+------ nL --- o 200 400 600 800 1000 Number of pierces Fig. 11 Schematic representation of the double

arcing, and the total number of the double arcing during 1000 times of cutting operation for the two types of nozzles.

Table 1 Nozzle specifications used in the cutting test.

	ΔL (mm)	D (mm)	d ₁ (mm)	d2 (mm)	dз (mm)
Proto	0.0	4.0	2.4	0.6	3.4
G940	1.0	2.0	0.0	1.5	3.4

method to determine an optimized geometry work shows a superior performance in arc stabilization inside the torch to the conventional α torch. Also, the present plasma torch can be operated with sufficiently high sp energy, ans has remarkable effects on the stable dross-free cutting. The nozzle shape obtained by the present

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